## OBISMS



Momentum will be conserved in two dimensions. The fuel was ejected in the y direction as seen from the ground, and so the fuel had no x-component of velocity.

$$p_x: \quad m_{\text{rocket}} v_0 = \left(m_{\text{rocket}} - m_{\text{fuel}}\right) v_x' + m_{\text{fuel}} 0 = \frac{2}{3} m_{\text{rocket}} v_x' \quad \rightarrow \quad \boxed{v_x' = \frac{3}{2} v_0}$$

$$p_{y}: 0 = m_{\text{fuel}} v_{\text{fuel}} + (m_{\text{rocket}} - m_{\text{fuel}}) v_{y}' = \frac{1}{3} m_{\text{rocket}} (2v_{0}) + \frac{2}{3} m_{\text{rocket}} v_{y}' \rightarrow v_{y}' = -v_{0}$$



We find the diameter of the spot from

$$\theta = \frac{\text{diameter}}{r_{\text{Earth-Moon}}} \rightarrow \text{diameter} = \theta r_{\text{Earth-Moon}} = (1.4 \times 10^{-5} \, \text{rad})(3.8 \times 10^8 \, \text{m}) = \boxed{5.3 \times 10^3 \, \text{m}}$$



(a) 1. The angular acceleration is related to the initial and final angular velocities:

$$\omega = \omega_0 + \alpha t = 0 + \alpha t$$

2. Solve for α:

$$\alpha = \frac{\omega}{t} = \frac{500 \text{ rev/min}}{5.5 \text{ s}} \times \frac{2\pi \text{ rad}}{1 \text{ rev}} \times \frac{1 \text{ min}}{60 \text{ s}}$$
$$= 9.52 \text{ rad/s}^2 = \boxed{9.5 \text{ rad/s}^2}$$

(b) 1. The angular displacement is related to the time by Equation 9-6:

$$\theta - \theta_0 = \omega_0 t + \frac{1}{2}\alpha t^2 = 0 + \frac{1}{2}(9.52 \text{ rad/s}^2)(5.5 \text{ s})^2$$
  
= 144 rad

2. Convert radians to revolutions:

$$144 \, \text{rad} \times \frac{1 \, \text{rev}}{2\pi \, \text{rad}} = 22.9 \, \text{rev} = \boxed{23 \, \text{rev}}$$



The interaction between the planet and the spacecraft is elastic, because the force of gravity is conservative. Thus kinetic energy is conserved in the interaction. Consider the problem a 1dimensional collision, with A representing the spacecraft and B representing Saturn. Because the mass of Saturn is so much bigger than the mass of the spacecraft, Saturn's speed is not changed appreciably during the interaction. Use Eq. 7-7, with  $v_A = 10.4 \, \mathrm{km/s}$  and  $v_B = v_B' = -9.6 \, \mathrm{km/s}$ .

$$v_A - v_B = -v_A' + v_B'$$
  $\rightarrow v_A' = 2v_B - v_A = 2(-9.6 \text{ km/s}) - 10.4 \text{ km/s} = \frac{-29.6 \text{ km/s}}{-29.6 \text{ km/s}}$   
s there is almost a threefold increase in the

Thus there is almost a threefold increase in the spacecraft's speed.



- (a) 1. Apply the definition of moment of inertia  $I = \sum m_i r_i^2$  (Equation 9-11), where  $r_i$  is the radial distance from the rotation axis to the particle of mass  $m_i$ :
  - 2. The masses  $m_i$  and the distances  $r_i$  are given:
  - 3. Substitution gives the moment of inertia:
  - 4. Using Equation 9-12, solve for the kinetic energy:
- (b) 1. To find the kinetic energy of the ith particle, we must first find its speed:
  - 2. The particles are all moving in circles of radius a. Find the speed of each particle:
  - Substitute into the Part-(b) step-1 result:
  - 4. Each particle has the same kinetic energy. Sum the kinetic energies to get the total:
  - 5. Compare with the Part-(a) result:

$$I = \sum_i m_i r_i^2 = m_1 r_1^2 + m_2 r_2^2 + m_3 r_3^2 + m_4 r_4^2$$

$$m_1 = m_2 = m_3 = m_4 = m$$

$$r_1 = r_2 = r_3 = r_4 = a$$

$$I = ma^2 + ma^2 + ma^2 + ma^2 = 4ma^2$$

$$K = \frac{1}{2}I\omega^2 = \frac{1}{2}4ma^2\omega^2 = 2ma^2\omega^2$$

$$K_i = \frac{1}{2}m_i v_i^2$$

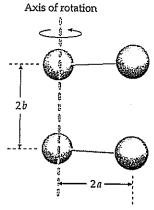
$$v_i = r_i \omega = a \omega \quad (i = 1, \dots, 4)$$

$$K_i = \frac{1}{2}m_i v_i^2 = \frac{1}{2}ma^2\omega^2$$

$$\begin{split} K &= \sum_{i=1}^4 K_i = \tfrac{1}{2} m_1 v_1^2 + \tfrac{1}{2} m_2 v_2^2 + \tfrac{1}{2} m_3 v_3^2 + \tfrac{1}{2} m_4 v_4^2 \\ &= 4 (\tfrac{1}{2} m a^2 \omega^2) = 2 m a^2 \omega^2 \end{split}$$

The two calculations give the same result.



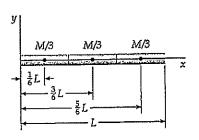




I= 5 mini= 2mp?

with R= ZQ = I= 8 en Q

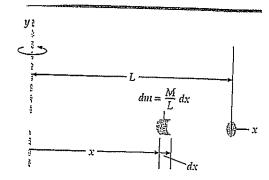




- . Sketch the rod divided into three segments and superpose the point-particle constructs at the center of each segment (Figure 9-5):
- FIGURE 9-5

- . Apply the equation  $I = \sum m_i r_i^2$  to the approximate system (the point-particle constructs):
- $I = \sum m_i r_i^2 \approx m_1 r_1^2 + m_2 r_2^2 + m_3 r_3^2$
- . The mass of each particle is  $\frac{1}{3}M$ , and the distances of the particles from the axis are  $\frac{1}{6}L$ ,  $\frac{3}{6}L$ , and  $\frac{5}{6}L$ :
- $I \approx (\frac{1}{3}M)(\frac{1}{6}L)^2 + (\frac{1}{3}M)(\frac{3}{6}L)^2 + (\frac{1}{3}M)(\frac{5}{6}L)^2$  $= \frac{1}{3}M\left(\frac{1+3^2+5^2}{6^2}\right)L^2 = \boxed{\frac{35}{108}ML^2}$





- 1. Draw a sketch (Figure 9-6) showing the rod along the +x axis with its end at the origin. To calculate I about the y axis, we choose a mass element dm at a distance x from the axis:
- 2. The moment of inertia about the y axis is given by the integral:
- $I = \int x^2 \, dm$
- 3. To compute the integral, we first relate dm to dx. Express dm in  $dm = \lambda dx = \frac{M}{L} dx$  terms of the linear mass density  $\lambda$  and dx:
- terms of the linear mass density λ and dx:
  4. Substitute and perform the integration. We choose integration limits so that we integrate through the mass distribution in the direction of increasing x:
- $I = \int x^{2} dm = \int_{0}^{L} x^{2} \frac{M}{L} dx = \frac{M}{L} \int_{0}^{L} x^{2} dx$  $= \frac{M}{L} \frac{1}{3} x^{3} \Big|_{0}^{L} = \frac{M}{L} \frac{L^{3}}{3} = \left[ \frac{1}{3} M L^{2} \right]$

FIGURE 9-6



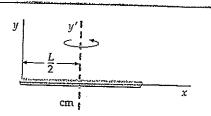


FIGURE 9-11

Cover the column to the right and try these on your own before looking at the answers.

## Steps

- 1. Apply the parallel-axis theorem to write I about the end in terms of  $I_{\rm cm}$ .
- Answers

$$I = I_{cm} + Mh^2$$
  
 $I_u = I_{u'} + M(\frac{1}{2}L)^2$ 

2. Substitute, using  $\frac{1}{3}ML^2$  for  $I_y,I_{\rm cm}$  for  $I_y$ , and solve for  $I_{\rm cm}$ .

$$I_{\rm cm} = I_y - Mh^2 = \frac{1}{3}ML^2 - M(\frac{1}{2}L)^2 = \sqrt{\frac{1}{12}ML^2}$$



The firing force of the rockets will create a net torque, but no net force. Since each rocket fires tangentially, each force has a lever arm equal to the radius of the satellite, and each force is perpendicular to the lever arm. Thus  $\tau_{\rm net}=4FR$ . This torque will cause an angular acceleration according to  $\tau=I\alpha$ , where  $I=\frac{1}{2}MR^2$  for a cylinder. The angular acceleration can be found from the kinematics by  $\alpha=\frac{\Delta\omega}{\Delta t}$ . Equating the two expressions for the torque and substituting enables us to solve for the force.

$$4FR = I\alpha = \frac{1}{2}MR^{2} \frac{\Delta\omega}{\Delta\tau}$$

$$F = \frac{MR\Delta\omega}{8\Delta t} = \frac{(3600 \text{ kg})(4.0 \text{ m})(32 \text{ rev/min})(2\pi \text{ rad/rev})(1 \text{ min/60 s})}{8(5.0 \text{ min})(60 \text{ s/min})}$$

$$= 20.11 \text{ N} \approx \boxed{2.0 \times 10^{1} \text{ N}}$$



(a) The moment of inertia of a thin rod, rotating about its end, is given in Figure 8-21(g). There are three blades to add.

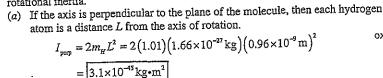
$$I_{\text{total}} = 3\left(\frac{1}{3}ML^2\right) = ML^2 = (160 \text{ kg})(3.75 \text{ m})^2 = 2250 \text{ kg} \cdot \text{m}^2 \approx 2.3 \times 10^2 \text{ kg} \cdot \text{m}^2$$

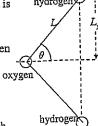
(b) The torque required is the rotational inertia times the angular acceleration, assumed constant.

$$\tau = I_{\text{total}} \alpha = I_{\text{total}} \frac{\omega - \omega_0}{t} = \left(2250 \,\text{kg} \cdot \text{m}^2\right) \frac{\left(5.0 \,\text{rev/sec}\right) \left(2\pi \,\text{rad/rev}\right)}{8.0 \,\text{s}} = \left[\frac{8.8 \times 10^3 \,\text{m} \cdot \text{N}}{10^3 \,\text{m}}\right]$$



The mass of a hydrogen atom is 1.01 atomic mass units. The atomic mass unit is  $1.66 \times 10^{-27} \, \mathrm{kg}$ . Since the axis passes through the oxygen atom, it will have no rotational inertia.

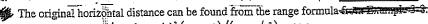




(b) If the axis is in the plane of the molecule, bisecting the H-O-H bonds, each hydrogen atom is a distance of  $L_y = L \sin \theta = (9.6 \times 10^{-10} \text{ m}) \sin 52^\circ$ 

= 
$$7.564 \times 10^{-10}$$
 m. Thus the moment of inertia is

$$I_{\text{plane}} = 2m_{y}L_{y}^{2} = 2(1.01)(1.66 \times 10^{-27} \text{kg})(7.564 \times 10^{-10} \text{m})^{2} = 1.9 \times 10^{-45} \text{kg} \cdot \text{m}^{2}$$



$$R = v_0^2 \sin 2\theta_0^2 / g = (25 \text{ m/s})^2 (\sin 60^\circ) / (9.8 \text{ m/s}^2) = 55.2 \text{ m}$$

The height at which the objects collide can be found  $v_{ij} = 0$  at the top of the path. Take up to be positive.

$$v_y^2 = v_{y0}^2 + 2a(y - y_0) \rightarrow (y - y_0) = \frac{v_y^2 - v_{y0}^2}{2a} = \frac{0 - \left[ (25 \text{ m/s}) \sin 30^\circ \right]^2}{2(-9.8 \text{ m/s}^2)} = 7.97 \text{ m}$$

Let m represent the bullet and M the skeet. When the objects collide, the skeet is moving horizontally at  $v_0 \cos \theta = (25 \, \text{m/s}) \cos 30^\circ = 21.65 \, \text{m/s} = v_x$ , and the bullet is moving vertically at  $v_y = 200 \, \text{m/s}$ . Write momentum conservation in both directions to find the velocities after the totally inelastic collision.

$$p_x: Mv_x = (M+m)v_x' \rightarrow v_x' = \frac{Mv_x}{M+m} = \frac{(0.25 \text{ kg})(21.65 \text{ m/s})}{(0.25+0.015) \text{ kg}} = 20.42 \text{ m/s}$$

$$p_y: mv_y = (M+m)v_y' \rightarrow v_y' = \frac{mv_y}{M+m} = \frac{(0.015 \text{ kg})(200 \text{ m/s})}{(0.25+0.015) \text{ kg}} = 11.32 \text{ m/s}$$

(a) The speed  $v_y'$  can be used as the starting vertical speed in the to find the height that the skeet-bullet combination rises above the point of collision.

$$v_y^2 = v_{y0}^2 + 2a(y - y_0)_{\text{extra}} \rightarrow (y - y_0)_{\text{extra}} = \frac{v_y^2 - v_{y0}^2}{2a} = \frac{0 - (11.32 \text{ m/s})^2}{2(-9.8 \text{ m/s}^2)} = \boxed{6.5 \text{ m}}$$

(b) From Eq. 2. The application the vertical motion after the collision, we can find the time for the skeet-bullet combination to reach the ground.

$$y = y_0 + v_y't + \frac{1}{2}at^2 \rightarrow 0 = 7.97 \text{ m} + (11.32 \text{ m/s})t + \frac{1}{2}(-9.8 \text{ m/s}^2)t^2 \rightarrow 4.9t^2 - 11.32t - 7.97 = 0 \rightarrow t = 2.88 \text{ s}, -0.565 \text{ s}$$

The positive time root is used to find the horizontal distance traveled by the combination after the collision.

$$x_{\text{after}} = v_x' t = (20.42 \text{ m/s})(2.88 \text{ s}) = 58.7 \text{ m}$$

If the collision would not have happened, the skeet would have gone  $\frac{1}{2}R$  horizontally.

$$\Delta x = x_{\text{after}} - \frac{1}{2}R = 58.7 \text{ m} - \frac{1}{2}(55.2 \text{ m}) = 31.1 \text{ m} \approx \boxed{31 \text{ m}}$$



See Slides